



✉ Box 848 – SE-98 128 Kiruna - Sweden

Czech Technical University  
Faculty of Electrical Engineering  
Department of Control Engineering  
Examination board

CTU Diploma Project Review  
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Division of Space Technology  
Department of Computer Science, Electrical  
and Space Engineering  
Luleå University of Technology

Phone +46 980 79100  
Fax +46 980 79190  
Email anita.enmark@ltu.se

CTU Diploma Project review- 2<sup>nd</sup> reviewer's evaluation of master thesis with title "Quadcopter flight mechanics model and control algorithms laws " by Space Master student Eswarmurthi Gopalakrishnan.

I find that the *goal* of the thesis project fulfils the requirements of a master thesis in space technology. The work concerns development a quadcopter flight mechanics nonlinear model in MATLAB/Simulink. A set of both basic and advanced control laws for stabilization and guidance should be validated for the model. Also, the thesis shall include a literature survey related to the thesis topic.

A rich source of documents concerning quadcopter modelling and control exists, many of which stem from amateurs and not from research project. Some of these documents are nevertheless very informative and in some cases also advanced. For the litterature survey it is of importance to include trustworthy litterature and present this in a consistent way for the reader.

Only a few sources are included for each "submodel" in the thesis. Mathematical models are presented more or less word by word from these sources and not in a consistent way. There is no part that really shows how to put together all "submodels" into a final full model. The sensor models are assumed to be ideal, i.e no noise or dynamics, and this is not commented on. There are also other important assumptions that are not commented on, for example the basic assumptions leading to a SISO controller approach with no crosscouplings ( $I_{xy}$ ,  $I_{yz}$  etc).

The control laws and models are only "validated" by a sanity check using visualisation. The transfer functions and blockdiagrams for the systems are generally not explicitly given and the results are not compared to analytical solutions for the linear or linearized cases. The final value theorem could for example have been used in order to validate the 3 degree steady state error mentioned on p.54.

It is not always clear if transfer functions are from reference to output or from disturbances to output (noise transfer functions are not considered at all). The discussions regarding the presented results (graphs) are shallow and results that should be discussed (for example fig 10 relation between angular velocity and position) are not commented on.

The controllers are designed ad hoc (probably by manually varying model controller paramerters) but goodnes parameters are not evaluated, i.e does the system meet given design constraints.

The thesis as a whole is of low quality and in order to approve the thesis with grade E the student has to complement the written thesis during the oral presentation and defence and fill in the shortcomings.

This review serves solely for the purposes of the diploma project defense at CTU. LTU official evaluation for the SpaceMaster double degree will follow the thesis defense and may differ from this review report and suggested grade.

Dr. Anita Enmark  
Luleå University of Technology